

Orbits in the problem of two fixed centers on the sphere

M.A. Gonzalez Leon¹, J. Mateos Guilarte² and M. de la Torre Mayado²

¹*Departamento de Matemática Aplicada, University of Salamanca, Spain*

²*Departamento de Física Fundamental, University of Salamanca, Spain*

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Abstract

A trajectory isomorphism between the two Newtonian fixed center problem in the sphere and two associated planar two center problems is constructed. The complete set of orbits in S^2 for this problem is calculated.

1 Introduction

The two fixed center problem on the two-dimensional sphere goes back to Killing [1], and in modern times to Kozlov and Harin [2], who proved the separability of the problem, thus its integrability, in sphero-conical coordinates. These coordinates on S^2 were introduced by Neumann [3] in one of the first examples of dynamics in spaces of constant curvature and they are closely related to elliptic coordinates, in fact the first system of coordinates plays a rôle in the dynamics on the sphere completely similar to the second system with respect to the Euclidian case. Integrability and Hamilton-Jacobi separability in sphero-conical coordinates has been constructed for different physical systems defined on the sphere, see for instance [4]. In particular, the authors analyzed in this context the Neumann problem and the Garnier system on S^2 in order to study solitary waves in one-dimensional non-linear S^2 -sigma models, see [5] and [6]. A detailed historical review of several systems defined in spaces of constant curvature, including open problems, has been recently performed in [7] where a precise bibliography is contained.

The two fixed center problem on the sphere is the superposition of two Kepler problems on S^2 . An explicit expression for the second constant of motion for this problem and also for some generalizations was given in [8, 9]. In [10] Borisov and Mamaev, inspired in a previous work of Albouy and Stuchi [11, 12], established a trajectory isomorphism (in terms of a new time variable) between the orbits lying in the half-sphere that contains the two attractive centers and the bounded orbits of an associated planar system of two attractive centers.

In this work we extend this result to the whole sphere, i.e. we establish a trajectory isomorphism between the complete set of orbits of the original problem and the corresponding one to two associated planar problems. The underlying idea is to identify each trajectory crossing the equator with the conjunction of two planar unbounded orbits, one of the two attractive center problem and another one for the system of two repulsive centers.

This extended trajectory isomorphism allows us to describe the bifurcation diagram of the spherical problem, analyzed in [13, 14, 15], in terms of the well known bifurcation diagrams for the planar problems (see [16] and [17]) and also determines a simple change of variables, from sphero-conical coordinates to planar elliptic coordinates, that converts the involved quadratures into elliptic integrals. Thus finally explicit formulas for the different types of orbits of the complete problem are obtained in terms of Jacobi elliptic functions both for the “radial” and “angular” coordinates. The existence of closed orbits in the sphere is guaranteed for the case of commensurability between the involved periods.

The structure of the paper is as follows: The problem is presented in Section 2 using sphero-conical coordinates on S^2 . In Section 3 the extended trajectory isomorphism is defined, and thus the quadratures are converted into elliptic integrals. The bifurcation diagram for the spherical problem is constructed from the diagrams of the two associated planar problems in Section 4. Finally, in Section 5, the processes of inversion of elliptic integrals in S^2 are detailed, and general equations for the solutions are showed.

The complete list of parametric equations for the different types of orbits in S^2 , in terms of a local time, is included in the Appendix, together with a gallery of figures for all the significative cases.

2 The two Newtonian centers problem in S^2

We consider the problem of a unit mass lying on the sphere S^2 of radius R , viewed as immersed in the Euclidean space \mathbb{R}^3 with cartesian coordinates (X, Y, Z) :

$$X^2 + Y^2 + Z^2 = R^2$$

under the influence of the superposition of two Kepler potentials on S^2 , i.e. the potential:

$$\mathcal{U}(\theta_1, \theta_2) = -\frac{\gamma_1}{R} \cotan \theta_1 - \frac{\gamma_2}{R} \cotan \theta_2 \quad (1)$$

where θ_1 and θ_2 denote the great circle angles between the location of the centers F_1 and F_2 , see Fig. 1, and a given point P on S^2 , in such a way that $R\theta_1$ and $R\theta_2$ are the orthodromic distances from F_1 and F_2 to P , respectively. γ_1 and γ_2 are the strengths of the centers, where we have considered $0 < \gamma_2 \leq \gamma_1$, i.e. the test mass feels the presence of two attractive centers in F_1 and F_2 , and correspondingly two repulsive centers in their antipodal points \bar{F}_1 and \bar{F}_2 . Without loss of generality, it has been chosen the points, notation and orientation showed in Fig. 1. Thus cartesian coordinates of F_1 and F_2 are: $(R \sin \theta_f, 0, R \cos \theta_f) = (R\bar{\sigma}, 0, R\sigma)$ and $(-R \sin \theta_f, 0, R \cos \theta_f) = (-R\bar{\sigma}, 0, R\sigma)$ respectively. Parameters $\sigma = \cos \theta_f$ and $\bar{\sigma} = \sin \theta_f$ have been introduced in order to alleviate the notation.

This problem is completely integrable, see e.g. [1, 2], there exist two constants of motion, the Hamiltonian:

$$\mathcal{H} = \frac{1}{2R^2} (L_X^2 + L_Y^2 + L_Z^2) - \frac{1}{R} \left(\frac{\gamma_1(\sigma Z + \bar{\sigma} X)}{\sqrt{R^2 - (\sigma Z + \bar{\sigma} X)^2}} + \frac{\gamma_2(\sigma Z - \bar{\sigma} X)}{\sqrt{R^2 - (\sigma Z - \bar{\sigma} X)^2}} \right) \quad (2)$$

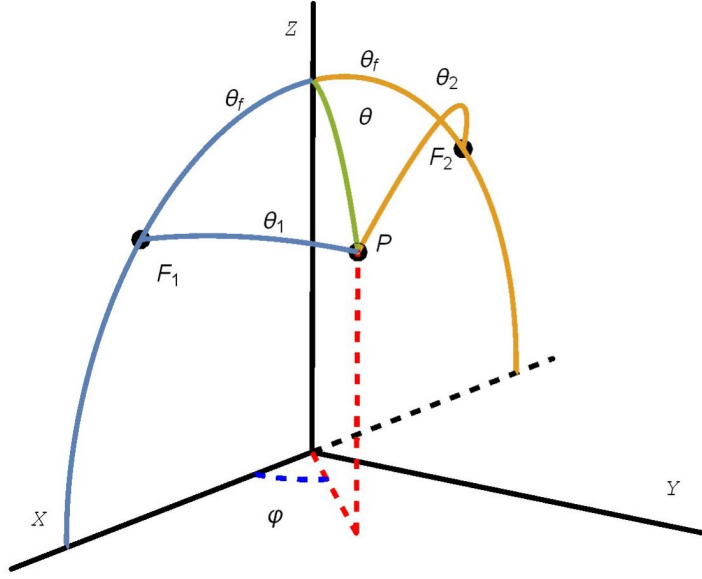


Figure 1: Location of the two Newtonian centers F_1 and F_2 in S^2 . The angular separation is $2\theta_f$, with $0 < \theta_f < \frac{\pi}{2}$. θ_1 and θ_2 denote the great circle angles between a given point $P \in S^2$ and F_1 and F_2 , respectively.

where: $\vec{L} = \vec{X} \times \vec{P}$, $\vec{P} = (P_X, P_Y, P_Z)$, $\vec{X} = (X, Y, Z)$, and the second invariant:

$$\Omega = \frac{1}{2R^2} (L_X^2 + \sigma^2 L_Y^2) - \frac{\sigma}{R} \left(\frac{\gamma_1 Z}{\sqrt{R^2 - (\sigma Z + \bar{\sigma} X)^2}} + \frac{\gamma_2 Z}{\sqrt{R^2 - (\sigma Z - \bar{\sigma} X)^2}} \right) \quad (3)$$

This constant of motion is slightly different but equivalent to the invariant obtained by Borisov and Mamaev in [8, 9]. Potential (1) can be rewritten as:

$$U(\theta_1, \theta_2) = - \frac{(\gamma_1 + \gamma_2) \sin \frac{\theta_1 + \theta_2}{2} \cos \frac{\theta_1 + \theta_2}{2} + (\gamma_1 - \gamma_2) \sin \frac{\theta_2 - \theta_1}{2} \cos \frac{\theta_2 - \theta_1}{2}}{R \left(\sin^2 \frac{\theta_1 + \theta_2}{2} - \sin^2 \frac{\theta_2 - \theta_1}{2} \right)} \quad (4)$$

in such a way that it is natural to introduce an *à la Euler* version of sphero-conical coordinates on S^2 , i.e.

$$U = \sin \frac{\theta_1 + \theta_2}{2}, \quad V = \sin \frac{\theta_2 - \theta_1}{2}; \quad -\bar{\sigma} < V < \bar{\sigma}, \quad \bar{\sigma} < U < 1$$

Coordinate lines with fixed U or V resemble “spherical ellipses” or “spherical hyperbolas” respectively with foci F_1 and F_2 , well understood that “spherical hyperbolas” are no more than “spherical ellipses” with respect to the pair of foci \bar{F}_1 and F_2 or F_1 and \bar{F}_2 .

The change of coordinates:

$$X = \frac{R}{\bar{\sigma}} UV, \quad Y^2 = \frac{R^2}{\sigma^2 \bar{\sigma}^2} (U^2 - \bar{\sigma}^2) (\bar{\sigma}^2 - V^2), \quad Z^2 = \frac{R^2}{\sigma^2} (1 - U^2) (1 - V^2) \quad (5)$$

is a four-to-one map because the ambiguities in the signs of Y and Z . Obviously coordinates U and V are dimensionless.

Potential (4) is written in these sphero-conical coordinates with two different expressions depending on the hemisphere that it is considered. For $S_+^2 = \{(X, Y, Z) \in S^2, Z \geq 0\}$, we have:

$$\mathcal{U}_+(U, V) = -\frac{1}{R(U^2 - V^2)} \left((\gamma_1 + \gamma_2)U\sqrt{1 - U^2} + (\gamma_1 - \gamma_2)V\sqrt{1 - V^2} \right)$$

whereas in $S_-^2 = \{(X, Y, Z) \in S^2, Z \leq 0\}$ the potential reads:

$$\mathcal{U}_-(U, V) = -\frac{1}{R(U^2 - V^2)} \left(-(\gamma_1 + \gamma_2)U\sqrt{1 - U^2} + (\gamma_1 - \gamma_2)V\sqrt{1 - V^2} \right)$$

Thus Hamiltonian (2) has also to be splitted in two different expressions:

$$\mathcal{H}_\pm = \frac{1}{2R^2(U^2 - V^2)} \left((U^2 - \bar{\sigma}^2)(1 - U^2)p_U^2 + (\bar{\sigma}^2 - V^2)(1 - V^2)p_V^2 \right) + \mathcal{U}_\pm(U, V) \quad (6)$$

The Hamilton-Jacobi equations coming from (6):

$$\mathcal{H}_\pm \left(\frac{\partial S}{\partial U}, \frac{\partial S}{\partial V}, U, V \right) + \frac{\partial S}{\partial t} = 0 \quad (7)$$

are separable into two ordinary differential equations if we look for solutions of the form: $S_\pm(t; U, V) = S_t(t) + S_{U\pm}(U) + S_V(V)$. Introducing nondimensional variables:

$$\mathcal{H}_\pm \rightarrow \frac{\gamma_1 + \gamma_2}{R} \mathcal{H}_\pm, \quad t \rightarrow \frac{\sqrt{R^3}}{\sqrt{\gamma_1 + \gamma_2}} t, \quad p_{U,V} \rightarrow \sqrt{R(\gamma_1 + \gamma_2)} p_{U,V}$$

and defining the parameter:

$$\gamma = \frac{\gamma_2}{\gamma_1 + \gamma_2}$$

the complete solution of (7) is:

$$\begin{aligned} S_\pm(t; U, V) = & -Ht + \text{sg}(p_U) \sqrt{2} \int_{\bar{\sigma}}^U \frac{\sqrt{HU^2 \pm U\sqrt{1 - U^2} - G}}{\sqrt{(1 - U^2)(U^2 - \bar{\sigma}^2)}} dU \\ & + \text{sg}(p_V) \sqrt{2} \int_{-\bar{\sigma}}^V \frac{\sqrt{-HV^2 + (1 - 2\gamma)V\sqrt{1 - V^2} + G}}{\sqrt{(\bar{\sigma}^2 - V^2)(1 - V^2)}} dV \end{aligned}$$

where H and G are the values of the constants of motion: $\mathcal{H} = H$, $\mathcal{G} = G$; \mathcal{G} is the separation constant, related with Ω and \mathcal{H} , (3) and (2), by the expression:

$$\mathcal{G} = \mathcal{H} - \Omega$$

Given the local time ς by: $d\varsigma = \frac{dt}{U^2 - V^2}$, the standard separation procedure leads us to the first order equations:

$$\frac{dU}{d\varsigma} = \text{sg}(p_U) \sqrt{2} \sqrt{(1 - U^2)(U^2 - \bar{\sigma}^2)(HU^2 + U\sqrt{1 - U^2} - G)} \quad (8)$$

$$\frac{dV}{d\varsigma} = \text{sg}(p_V) \sqrt{2} \sqrt{(1 - V^2)(\bar{\sigma}^2 - V^2)(-HV^2 + (1 - 2\gamma)V\sqrt{1 - V^2} + G)} \quad (9)$$

for the problem in the Northern hemisphere S_+^2 , and:

$$\frac{dU}{d\zeta} = \text{sg}(p_U) \sqrt{2} \sqrt{(1-U^2)(U^2-\bar{\sigma}^2)(HU^2-U\sqrt{1-U^2}-G)} \quad (10)$$

$$\frac{dV}{d\zeta} = \text{sg}(p_V) \sqrt{2} \sqrt{(1-V^2)(\bar{\sigma}^2-V^2)(-HV^2+(1-2\gamma)V\sqrt{1-V^2}+G)} \quad (11)$$

for the Southern S_-^2 one.

A direct attack to the involved quadratures looks apparently cumbersome, and as far as we know they are not solved in the literature. Nevertheless some of the qualitative and topological properties of these orbits have been analyzed in [13, 14, 15].

3 Trajectory isomorphism between the spherical and two different planar problems

Following Borisov & Mamaev [10] we go back to Cartesian coordinates (X, Y, Z) where the potential (1) can be written as:

$$\mathcal{U}(X, Y, Z) = -\frac{1}{R} \left(\frac{\gamma_1(\sigma Z + \bar{\sigma} X)}{\sqrt{R^2 - (\sigma Z + \bar{\sigma} X)^2}} + \frac{\gamma_2(\sigma Z - \bar{\sigma} X)}{\sqrt{R^2 - (\sigma Z - \bar{\sigma} X)^2}} \right) \quad (12)$$

The corresponding Newton equations for this problem are:

$$\ddot{X} = -\frac{\partial \mathcal{U}}{\partial X} + \lambda X, \quad \ddot{Y} = -\frac{\partial \mathcal{U}}{\partial Y} + \lambda Y, \quad \ddot{Z} = -\frac{\partial \mathcal{U}}{\partial Z} + \lambda Z \quad (13)$$

where dots represent derivatives with respect to the physical (dimensional) time t and λ is the Lagrange multiplier. In [10] it was proved that the gnomonic projection from S_+^2 to the tangent plane Π_+ at the point $(0, 0, R)$, together with a linear transformation in Π_+ , maps Newton equations (13) to the Newton equations of an associated problem of two attractive centers in \mathbb{R}^2 .

Here, we shall also consider simultaneously another gnomonic projection, from S_-^2 to the tangent plane Π_- , at $(0, 0, -R)$. The projected coordinates (x, y) are given in the two planes by:

$$\Pi_+ : \quad x = \frac{R}{Z} X, \quad y = \frac{R}{Z} Y \quad ; \quad \Pi_- : \quad x = \frac{R}{-Z} X, \quad y = \frac{R}{-Z} Y \quad (14)$$

We will use throughout the paper the following criteria: uppercase letters describe magnitudes and variables specifically defined in the sphere, whereas lowercase will be associated to the planar cases. Following [10] we perform in Π_+ the linear transformation:

$$x_1 \equiv x, \quad x_2 \equiv \frac{y}{\sigma} \quad (15)$$

Newton equations (13) for potential (12) are re-written in transformed projected coordinates (x_1, x_2) on Π_+ as:

$$x_1''(\tau) = -\frac{\partial \mathcal{V}_+}{\partial x_1}, \quad x_2''(\tau) = -\frac{\partial \mathcal{V}_+}{\partial x_2} \quad (16)$$

$$\mathcal{V}_+(x_1, x_2) = -\frac{\alpha_1}{\sqrt{(x_1 - a)^2 + x_2^2}} - \frac{\alpha_2}{\sqrt{(x_1 + a)^2 + x_2^2}} \quad (17)$$

where primes denote derivative with respect to a new time τ defined by:

$$d\tau = \frac{R^2}{Z^2} dt$$

and we have introduced the parameters: $a = R\frac{\sigma}{\sigma}$, $\alpha_1 = \frac{\gamma_1}{\sigma^2}$ and $\alpha_2 = \frac{\gamma_2}{\sigma^2}$.

Simili modo, Newton equations (13) restricted to S_-^2 can be projected into Π_- using (14) and, after applying transformation (15), the equations:

$$\begin{aligned} x_1''(\tau) &= -\frac{\partial \mathcal{V}_-}{\partial x_1}, & x_2''(\tau) &= -\frac{\partial \mathcal{V}_-}{\partial x_2} \\ \mathcal{V}_-(x_1, x_2) &= \frac{\alpha_2}{\sqrt{(x_1 - a)^2 + x_2^2}} + \frac{\alpha_1}{\sqrt{(x_1 + a)^2 + x_2^2}} \end{aligned} \quad (18)$$

are obtained.

Note that $\mathcal{V}_-(x_1, x_2)$ in Π_- is no more than the planar potential of two repulsive centers, where the rôles of the points $(\pm a, 0)$, and thus the strengths of the centers in modulus, are interchanged with respect to the attractive potential $\mathcal{V}_+(x_1, x_2)$ in Π_+ .

Thus, while the restriction of Newton equations (13) to the Northern hemisphere S_+^2 is equivalent to the Newton equations (16) for a planar problem of two attractive centers with potential (17), the restriction to the Southern hemisphere S_-^2 is tantamount to a planar problem of two repulsive centers with potential (18).

Bounded orbits of the attractive planar problem are in a one-to-one correspondence with the orbits of the spherical problem that lie in S_+^2 . However, trajectories of the spherical problem crossing the equator have to be described in this projected picture by two pieces: an unbounded orbit of the attractive planar problem (17) in Π_+ plus an (unbounded) orbit of the repulsive planar problem (18) in Π_- , corresponding to the parts of the orbit belonging to S_+^2 and S_-^2 respectively.

It is possible to describe in a compact form the two associated planar problems, in Π_+ and Π_- respectively, by the hamiltonians:

$$h_{\pm} = \frac{1}{2} (p_1^2 + p_2^2) + \mathcal{V}_{\pm}(x_1, x_2) \quad (19)$$

It is adequate again to use non-dimensional variables:

$$x_i \rightarrow ax_i, \quad p_i \rightarrow \frac{\sqrt{\alpha_1 + \alpha_2}}{\sqrt{a}} p_i, \quad \tau \rightarrow \frac{\sqrt{a^3}}{\sqrt{\alpha_1 + \alpha_2}} \tau, \quad h_{\pm} = \frac{\alpha_1 + \alpha_2}{a} h_{\pm}; \quad \alpha = \frac{\alpha_2}{\alpha_1 + \alpha_2} = \gamma$$

and to introduce ‘‘radial’’, u and ‘‘angular’’, v elliptic (Euler) coordinates in \mathbb{R}^2 :

$$u = \frac{\sqrt{(x_1 + 1)^2 + x_2^2} + \sqrt{(x_1 - 1)^2 + x_2^2}}{2}, \quad v = \frac{\sqrt{(x_1 + 1)^2 + x_2^2} - \sqrt{(x_1 - 1)^2 + x_2^2}}{2}$$

$$x_1 = uv, \quad x_2 = \pm \sqrt{u^2 - 1} \sqrt{1 - v^2}, \quad v \in (-1, 1), \quad u > 1$$

in such a way that the hamiltonians (19) are written in terms of these coordinates as:

$$h_{\pm} = \frac{1}{u^2 - v^2} \left(\frac{u^2 - 1}{2} p_u^2 \mp u + \frac{1 - v^2}{2} p_v^2 - (1 - 2\alpha)v \right)$$

i.e. two standard Liouville-separable systems in elliptic coordinates. It is straightforward to construct the associated first order equations with respect to the local time $\zeta = \zeta(\tau)$ defined by:

$$d\zeta = \frac{d\tau}{u^2 - v^2}$$

and we finally obtain the following equations in the Π_+ plane:

$$\left(\frac{du}{d\zeta}\right)^2 = 2(u^2 - 1)(hu^2 + u - g), \quad \left(\frac{dv}{d\zeta}\right)^2 = 2(1 - v^2)(-hv^2 + (1 - 2\alpha)v + g) \quad (20)$$

that solve the original problem in S_+^2 . Correspondingly, for the Southern case we obtain in the Π_- plane:

$$\left(\frac{du}{d\zeta}\right)^2 = 2(u^2 - 1)(\tilde{h}u^2 - u - \tilde{g}), \quad \left(\frac{dv}{d\zeta}\right)^2 = 2(1 - v^2)(-\tilde{h}v^2 + (1 - 2\alpha)v + \tilde{g}) \quad (21)$$

where the constants of motion take the values: $h_+ = h$ and $g_+ = g$ for the energy and the separation constant in Π_+ , respectively; and $h_- = \tilde{h}$ and $g_- = \tilde{g}$ in Π_- . The quadratures involved in equations (20) and (21) are of elliptic type, and thus expressible in terms of the Jacobi elliptic functions.

It is possible to synthesize the chain of maps leading from the original problem in the sphere to the pair of planar two center problems (20) and (21) in a unique one-to-one transformation of coordinates in S^2 , from sphero-conical (U, V) to planar elliptic (u, v) , as follows:

$$U = \frac{\bar{\sigma}u}{\sqrt{\bar{\sigma}^2u^2 + \sigma^2}}; \quad V = \frac{\bar{\sigma}v}{\sqrt{\bar{\sigma}^2v^2 + \sigma^2}} \quad (22)$$

together with an equivalence, up to a constant factor, between the nondimensional local time ς of the spherical problem and the nondimensional local time ζ for the associated planar problems:

$$d\varsigma = \sqrt{\sigma\bar{\sigma}} d\zeta \quad (23)$$

The equator $Z = 0$, or $U = 1$, of S^2 is mapped by (22) into the point of infinity in the coordinate u .

Thus (22) and (23) map directly the first order equations (8, 9) in S_+^2 to equations (20) in Π_+ , and (10, 11) in S_-^2 to (21) in Π_- via the identifications:

$$h = \frac{\bar{\sigma}}{\sigma}(H - G) = \frac{\bar{\sigma}}{\sigma}\Omega = \tan\theta_f \Omega, \quad g = \frac{\sigma}{\bar{\sigma}}G = \cotan\theta_f G, \quad \text{in } S_+^2$$

$$\tilde{h} = \frac{\bar{\sigma}}{\sigma}(H - G) = \frac{\bar{\sigma}}{\sigma}\Omega = \tan\theta_f \Omega, \quad \tilde{g} = \frac{\sigma}{\bar{\sigma}}G = \cotan\theta_f G, \quad \text{in } S_-^2$$

It is remarkable that in this projected picture the rôle of the planar energies h and \tilde{h} is played, up to a factor, by the projection of the second constant of motion Ω , and not by the projection of the spherical Hamiltonian.

Consequently the transformation (22) establishes that fixing in S^2 a negative value of the constant of motion Ω , the orbits of the problem lie in the S_+^2 hemisphere and are in a one-to-one correspondence with the bounded orbits, $h = \frac{\bar{\sigma}}{\sigma}\Omega < 0$, of the planar attractive system in the Π_+ plane. However, if $\Omega \geq 0$, orbits cross the equator of S^2 , and thus the portions of the orbits belonging to S_+^2 are described by equations (20) with planar energy $h \geq 0$, unbounded planar orbits in the attractive problem in Π_+ , whereas the portions lying in the Southern hemisphere S_-^2 are determined by equations (21) with $\tilde{h} > 0$, i.e. unbounded planar orbits of the repulsive problem in Π_- .

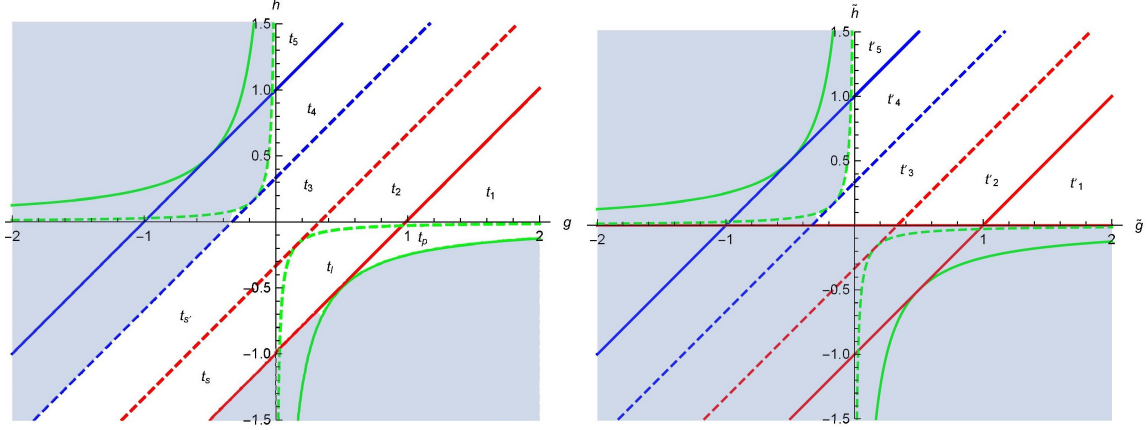


Figure 2: a) Bifurcation diagram for two attractive centers in the plane. b) Bifurcation diagram for two repulsive centers in the plane with the strengths (in modulus) exchanged with respect to the attractive potential. In both cases we chose $\alpha = \frac{1}{3}$.

4 The bifurcation diagrams

The isomorphic transformation (22) allows us to analyze the bifurcation diagram in S^2 starting from the bifurcation diagrams of the two associated planar problems. We shall reproduce the results explained in [14, 15] about the spherical problem constructing a global bifurcation diagram out of the diagrams of two planar centers, see [16] and [17], respectively attractive in Π_+ and repulsive in Π_- and strengths interchanged.

Both in Π_+ and Π_- planes, i.e. the images of the North S^2_+ and South S^2_- hemispheres, we re-write (20) and (21) in terms of the ramification points:

$$\left(\frac{du}{d\zeta}\right)^2 = 2h(u^2 - 1)(u - u_1)(u - u_2), \quad \left(\frac{dv}{d\zeta}\right)^2 = -2h(1 - v^2)(v - v_1)(v - v_2) \quad (24)$$

$$\Pi_+ : \quad u_1 = \frac{-1}{2h} - \sqrt{\frac{g}{h} + \frac{1}{4h^2}}, \quad u_2 = \frac{-1}{2h} + \sqrt{\frac{g}{h} + \frac{1}{4h^2}}$$

$$v_1 = \frac{1 - 2\alpha}{2h} - \sqrt{\frac{g}{h} + \frac{(1 - 2\alpha)^2}{4h^2}}, \quad v_2 = \frac{1 - 2\alpha}{2h} + \sqrt{\frac{g}{h} + \frac{(1 - 2\alpha)^2}{4h^2}},$$

$$\left(\frac{d\tilde{u}}{d\tilde{\zeta}}\right)^2 = 2\tilde{h}(\tilde{u}^2 - 1)(\tilde{u} - \tilde{u}_1)(\tilde{u} - \tilde{u}_2), \quad \left(\frac{d\tilde{v}}{d\tilde{\zeta}}\right)^2 = -2\tilde{h}(1 - \tilde{v}^2)(\tilde{v} - \tilde{v}_1)(\tilde{v} - \tilde{v}_2) \quad (25)$$

$$\Pi_- : \quad \tilde{u}_1 = \frac{1}{2\tilde{h}} - \sqrt{\frac{\tilde{g}}{\tilde{h}} + \frac{1}{4\tilde{h}^2}}, \quad \tilde{u}_2 = \frac{1}{2\tilde{h}} + \sqrt{\frac{\tilde{g}}{\tilde{h}} + \frac{1}{4\tilde{h}^2}}$$

$$\tilde{v}_1 = \frac{1 - 2\alpha}{2\tilde{h}} - \sqrt{\frac{\tilde{g}}{\tilde{h}} + \frac{(1 - 2\alpha)^2}{4\tilde{h}^2}}, \quad \tilde{v}_2 = \frac{1 - 2\alpha}{2\tilde{h}} + \sqrt{\frac{\tilde{g}}{\tilde{h}} + \frac{(1 - 2\alpha)^2}{4\tilde{h}^2}}$$

In Figure 2, plotted for $\alpha = 1/3$, we observe the bifurcation diagrams corresponding to the attractive and repulsive planar problems in Π_+ Fig. 2a) and Π_- Fig. 2b), respectively, with strengths α_1, α_2 ,

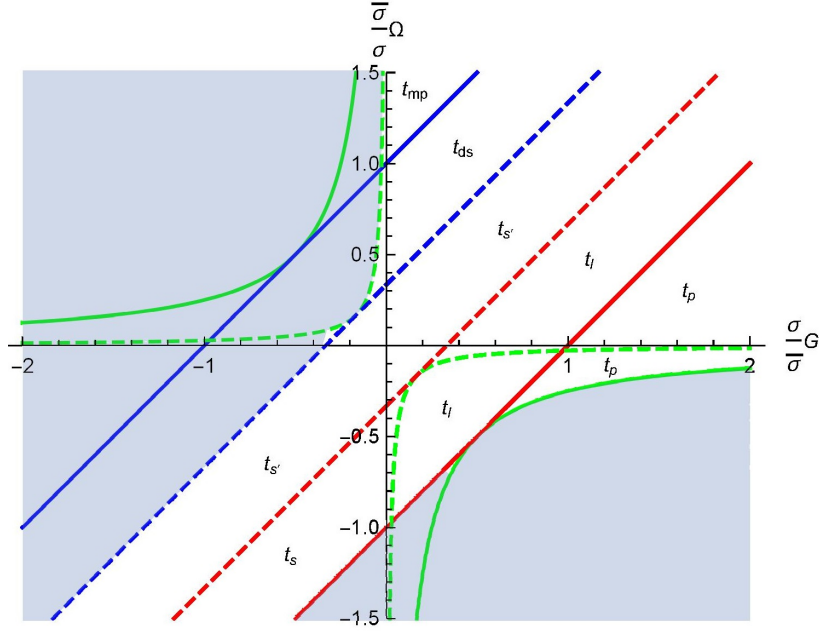


Figure 3: Global bifurcation diagram in S^2 with $\gamma = \frac{1}{3}$.

and $\tilde{\alpha}_1 = -\alpha_2$, $\tilde{\alpha}_2 = -\alpha_1$. Critical curves in both $\{h, g\}$ and $\{\tilde{h}, \tilde{g}\}$ planes are determined by the existence of double roots in (24) and (25), see [16, 17], and shadowed areas in the diagrams are zones where motion is classically forbidden, i.e., velocities and/or momenta are imaginary.

The allowed motions in the Π_+ -plane are of two types: (1) If $h < 0$ orbits are bounded and are usually labelled as $\{t_s, t_{s'}, t_l, t_p\}$, for satellitary, lemniscatic and planetary, see [16]. (2) If $h \geq 0$, see [17], unbounded orbits occur standardly labelled as: $\{t_1, t_2, t_3, t_4, t_5\}$. Separatrices between bounded and unbounded motions live in the $\{h = 0\}$ straight line.

In the Π_- plane a similar but simpler picture is found, see Fig. 2b). On the $\tilde{h} > 0$ upper half-plane unbounded orbits exist in five different classes, labeled as $\{t'_1, t'_2, t'_3, t'_4, t'_5\}$. In this case the line $\{\tilde{h} = 0\}$ does not accommodate separatrices but rather it responds to a limiting behaviour of unbounded zero energy orbits reached from $\tilde{h} > 0$.

The bifurcation diagram for the complete problem in S^2 , Figure 3, can now be constructed from the planar ones, using transformations (22) and (23). Two morphisms are induced: (1) Orbits in Π_+ are applied to orbits in S^2_+ identifying the invariants as follows: $h = \frac{\sigma}{\tilde{\sigma}}\Omega$ and $g = \frac{\sigma}{\tilde{\sigma}}G$. (2) Orbits in Π_- are applied to orbits in S^2_- if the invariants are translated to: $\tilde{h} = \frac{\sigma}{\tilde{\sigma}}\Omega$ and $\tilde{g} = \frac{\sigma}{\tilde{\sigma}}G$. The global bifurcation diagram in S^2 is thus displayed on the $\{\frac{\sigma}{\tilde{\sigma}}\Omega, \frac{\sigma}{\tilde{\sigma}}G\}$ -plane.

Moreover, the lower half plane of Figure 3, $\frac{\sigma}{\tilde{\sigma}}\Omega < 0$, is mapped one-to-one with the lower half plane of the problem of two attractive centers in Π_+ , Fig. 2a), as it was showed in [10], orbits lying only in S^2_+ are in a bijective correspondence with bounded orbits in Π_+ . However, fixed an initial condition, each point $(\frac{\sigma}{\tilde{\sigma}}\Omega, \frac{\sigma}{\tilde{\sigma}}G)$ in the upper half plane of the global diagram represents an orbit that crosses the equator of S^2 , and thus is mapped by (22) and (23) to the union of an unbounded orbit in Π_+ and another one in Π_- , with equal planar energies: $h = \tilde{h} = \frac{\sigma}{\tilde{\sigma}}\Omega$.

Critical curves in Figure 3 are inherited from the corresponding ones in planar diagrams:

- Double roots in equations (24, 25) for the “radial” variable arise in the: Blue straight line: $\mathcal{L}_1^2 = \{\frac{\bar{\sigma}}{\sigma}\Omega - \frac{\sigma}{\bar{\sigma}}G - 1 = 0\}$, red straight line: $\mathcal{L}_1^1 = \{\frac{\bar{\sigma}}{\sigma}\Omega - \frac{\sigma}{\bar{\sigma}}G + 1 = 0\}$, and green hyperbola: $\mathcal{L}_1^3 = \{4\Omega G + 1 = 0\}$.
- Analogously, double roots for the “angular” variable produce the: Dashed blue straight line: $\mathcal{L}_\gamma^1 = \{\frac{\bar{\sigma}}{\sigma}\Omega - \frac{\sigma}{\bar{\sigma}}G - (1 - 2\gamma) = 0\}$, dashed red straight line: $\mathcal{L}_\gamma^2 = \{\frac{\bar{\sigma}}{\sigma}\Omega - \frac{\sigma}{\bar{\sigma}}G + (1 - 2\gamma) = 0\}$, and dashed green hyperbola: $\mathcal{L}_\gamma^3 = \{4\Omega G + (1 - 2\gamma)^2 = 0\}$.

Orbits with $\Omega < 0$ are naturally labeled with the inherited standard notation for bounded motion in the planar associated problem in Π_+ . The branching points u_1, u_2 and v_1, v_2 , understood as functions of Ω and G , allow us to specify the analytical features of these orbits, in S_+^2 :

- Planetary orbits (t_p). There are two analytical possibilities that lead to the same type of orbits:

$$(1) \quad -1 < 1 < u_1 < u < u_2 \quad , \quad -1 < v < 1 \quad , \quad v_1, v_2 \in \mathbb{C} \quad (26)$$

$$(2) \quad -1 < 1 < u_1 < u < u_2 \quad , \quad v_1 < v_2 < -1 < v < 1 \quad (27)$$

In both cases the bounds $u = u_1$ and $u = u_2$ represent two caustics for these orbits, i.e. two “spherical ellipses” in the Northern hemisphere S_+^2 , see Fig. 6 (a), that confine the planetary motion of these “circumbinary” orbits.

- Lemniscatic orbits (t_l). Analogously, there exist two possibilities:

$$(1) \quad -1 < u_1 < 1 < u < u_2 \quad , \quad -1 < v < 1 \quad , \quad v_1, v_2 \in \mathbb{C} \quad (28)$$

$$(2) \quad -1 < u_1 < 1 < u < u_2 \quad , \quad v_1 < v_2 < -1 < v < 1 \quad (29)$$

A unique caustic, $u = u_2$, appears in this case. The orbits describe a lemniscatic motion around the two centers in S_2^+ . See Fig. 6 (b).

- Satellitary orbits (t_s): Each point $(\frac{\bar{\sigma}}{\sigma}\Omega, \frac{\sigma}{\bar{\sigma}}G)$ of this region in Figure 3 represents two possible orbits:

$$(1) \quad -1 < u_1 < 1 < u < u_2 \quad , \quad -1 < v_1 < v_2 < v < 1 \quad (30)$$

around the stronger center, limited by the caustics: $u = u_2$ and $v = v_2$, and:

$$(2) \quad -1 < u_1 < 1 < u < u_2 \quad , \quad -1 < v < v_1 < v_2 < 1 \quad (31)$$

around the weaker center, bounded by: $u = u_2$ and $v = v_1$. See Figure 6 (g).

- Satellitary orbits ($t_{s'}$) around the stronger center:

$$-1 < u_1 < 1 < u < u_2 \quad , \quad v_1 < -1 < v_2 < v < 1 \quad (32)$$

For this situation the motion is limited by the caustics: $u = u_2$ and $v = v_2$, see Figure 6 (c).

For $\Omega > 0$, it is possible to extend the standard nomenclator, Planetary (t_p), Lemniscatic (t_l) and Satellitary ($t_{s'}$), to the orbits that cross the equator but have a behavior analogous to the corresponding cases restricted to the Northern hemisphere. However, two completely new types of orbits arise. There are two zones of admissible motion without partners between the orbits with $\Omega < 0$, that we will call Dual Satellitary (t_{ds}) and Meridian Planetary (t_{mp}) orbits, taking into account its qualitative features.

Branching points are now identified by: $\tilde{u}_1 = -u_2$, $\tilde{u}_2 = -u_1$ and $\tilde{v}_1 = v_1$, $\tilde{v}_2 = v_2$, because: $h = \tilde{h} = \frac{\sigma}{\sigma'}\Omega$ and $g = \tilde{g} = \frac{\sigma}{\sigma'}G$ in order to glue continuously the two orbit pieces on the Northern and Southern hemispheres at the equator.

- Planetary orbits (t_p): The orbits in S^2 are composed by two pieces:

$$\begin{aligned} S_+^2 : & \quad u_1 < -1 < 1 < u_2 < u, & \quad v_1 < -1 < v < 1 < v_2 \\ S_-^2 : & \quad \tilde{u}_1 < -1 < 1 < \tilde{u}_2 < u, & \quad \tilde{v}_1 < -1 < v < 1 < \tilde{v}_2 \end{aligned} \quad (33)$$

Note that the limit $u \rightarrow \infty$ in both cases is no more that $U \rightarrow 1$, and thus the map (22) applies two unbounded curves to a finite one that crosses the equator of S^2 . The Northern pieces presents the caustic: $u = u_2$, whereas the Southern ones are limited by the ‘‘spherical ellipse’’: $u = \tilde{u}_2$. The motion is confined between these curves in a planetary way and can be seen as the natural continuation of the t_p orbits in S_+^2 with $\Omega < 0$. See Figure 6 (d).

- Lemniscatic orbits (t_l): Analogously, there are two parts:

$$\begin{aligned} S_+^2 : & \quad u_1 < -1 < u_2 < 1 < u, & \quad v_1 < -1 < v < 1 < v_2 \\ S_-^2 : & \quad -1 < \tilde{u}_1 < 1 < \tilde{u}_2 < u, & \quad \tilde{v}_1 < -1 < v < 1 < \tilde{v}_2 \end{aligned} \quad (34)$$

in such a way that there are no caustics in S_+^2 and one in S_-^2 : $u = \tilde{u}_2$. We find again a natural resemblance between these orbits and their partners in the $\Omega < 0$ case. See Figure 6 (e).

- Satellitary orbits ($t_{s'}$):

$$\begin{aligned} S_+^2 : & \quad u_1 < -1 < u_2 < 1 < u, & \quad -1 < v_1 < v < 1 < v_2 \\ S_-^2 : & \quad -1 < \tilde{u}_1 < 1 < \tilde{u}_2 < u, & \quad -1 < \tilde{v}_1 < v < 1 < \tilde{v}_2 \end{aligned} \quad (35)$$

The caustics are now: $u = \tilde{u}_2$ in S_-^2 , and $v = v_1 = \tilde{v}_1$ in the two hemispheres. See Figure 6 (f).

- Dual Satellitary orbits (t_{ds}):

$$\begin{aligned} S_+^2 : & \quad u_1 < -1 < u_2 < 1 < u, & \quad -1 < v_1 < v < v_2 < 1 \\ S_-^2 : & \quad -1 < \tilde{u}_1 < 1 < \tilde{u}_2 < u, & \quad -1 < \tilde{v}_1 < v < \tilde{v}_2 < 1 \end{aligned} \quad (36)$$

The t_{ds} orbits present a behaviour delimited by the two caustics: $v = v_1 = \tilde{v}_1$ and $v = v_2 = \tilde{v}_2$ in S^2 , and: $u = \tilde{u}_2$ in the Southern hemisphere. Thus the orbits pass between the two centers in S_+^2 , but do not reached the South Pole. See Figure 6 (h).

- Meridian Planetary orbits (t_{mp}):

$$\begin{aligned} S_+^2 : & \quad -1 < u_1 < u_2 < 1 < u, & \quad -1 < v_1 < v < v_2 < 1 \\ S_-^2 : & \quad -1 < \tilde{u}_1 < \tilde{u}_2 < 1 < u, & \quad -1 < \tilde{v}_1 < v < \tilde{v}_2 < 1 \end{aligned} \quad (37)$$

The situation is similar to the t_{ds} case, but now only the two “angular” caustics are allowable. Thus the orbits complete the passing between the centers not only in S_+^2 but also in S_-^2 . The t_{mp} orbits resemble the planetary ones interchanging the surrounded centers. See Figure 6 (i).

Finally, the analysis should be completed with the case $\Omega = 0$ whose orbits lie in the S_+^2 hemisphere. These can be easily described as the limit $\Omega \rightarrow 0$ in the $\Omega < 0$ case. The caustic $u = u_2$ for the t_p , t_l and $t_{s'}$ orbits becomes $u_2 \rightarrow \infty$, and thus $U(u_2) \rightarrow 1$, i.e. the equator $Z = 0$ of S^2 . Consequently the motions are completely similar to the corresponding ones in S_+^2 but now bounded by the equator.

5 Evaluation of the quadratures, inversion of the elliptic integrals

Explicit analytical expressions determining the orbits are obtained by applying standard procedures that require the inversion of elliptic integrals, see for instance [18, 19]. The quadratures solving the two pairs of uncoupled ODE’s (24) and (25) go back to Euler, Lagrange and Jacobi and have been thoroughly discussed by several authors along the time, see [20] and references therein, see also [21]. We shall briefly report here on the processes of quadrature evaluation/elliptic integral inversion in the context of the spherical problem, keeping in mind that the variables (u, v) , which appear in equations (24) and (25), should be regarded as coordinates in S_+^2 and S_-^2 through the map transformation (22), as it has been explained in the previous sections.

There are two distinctly different situations, for the $\Omega < 0$ or $\Omega > 0$ ranges:

- $\Omega < 0$. In this case the inversion of the elliptic integrals appearing in equations (24) is standard, we will detail only the planetary case as example.

The range for the u -variable in (24) (left) is: $u_1 < u < u_2$, and thus the curves: $u = u_1$ and $u = u_2$, $\forall v \in (-1, 1)$, determine the two caustics. The quadrature solving the u -equation in (24) is:

$$\pm \sqrt{-\frac{2\bar{\sigma}}{\sigma}} \Omega \zeta = I(u) - I(u_0); \quad I(u) = \int_{u_1}^u \frac{dz}{\sqrt{(z^2 - 1)(z - u_1)(u_2 - z)}} \quad (38)$$

where the initial condition $u(0) = u_0$ is assumed. The elliptic integral of the first kind $I(u)$ in (38) can be inverted performing the following change of variable $z \rightarrow s$, see [18] case 256:

$$z = \frac{u_1(1 - u_2) + (u_2 - u_1) \operatorname{sn}^2 s}{1 - u_2 + (u_2 - u_1) \operatorname{sn}^2 s} \quad \Rightarrow \quad I(u) = g_u \int_0^{s_u} ds = g_u s_u$$

where $\operatorname{sn} s$ denotes the Jacobi sinus function: $\operatorname{sn} s \equiv \operatorname{sn}(s|k_u^2)$, g_u and the elliptic modulus k_u , are defined in terms of the turning points as:

$$k_u^2 = \frac{2(u_2 - u_1)}{(u_2 - 1)(u_1 + 1)}, \quad g_u = \frac{2}{\sqrt{(u_2 - 1)(u_1 + 1)}}.$$

Formula (38) is thus simplified to become a linear relation between s_u and the local time ζ which is easily inverted:

$$g_u (s_u - s_{u_0}) = \pm \sqrt{-\frac{2\bar{\sigma}}{\sigma} \Omega} \zeta \quad \Rightarrow \quad s_u(\zeta) = \frac{\pm \sqrt{-\frac{2\bar{\sigma}}{\sigma} \Omega}}{g_u} \zeta + s_{u_0} \quad (39)$$

with: $g_u s_{u_0} = I(u_0)$. Finally, reminding the last change of variable, the explicit inversion of (38) is achieved:

$$u(\zeta) = \frac{u_1(1 - u_2) + (u_2 - u_1) \operatorname{sn}^2 s_u}{1 - u_2 + (u_2 - u_1) \operatorname{sn}^2 s_u}$$

where s_u is defined as function of the local time ζ , $s_u(\zeta)$, in equation (39). Alternatively, using the properties of Jacobi elliptic functions, $u(\zeta)$ can be re-written in terms of the Jacobi function dn in the simpler form:

$$u(\zeta) = \frac{u_1 - 1 + (u_1 + 1) \operatorname{dn}^2 s_u}{1 - u_1 + (u_1 + 1) \operatorname{dn}^2 s_u}, \quad -1 < 1 < u_1 < u < u_2. \quad (40)$$

We stress, by writing the inequalities characterizing this type of orbits, that the analytic expression for $u(\zeta)$ appearing in formula (40) is compelled to live inside the (u_1, u_2) interval.

The companion expression for $v(\zeta)$, for instance in the planetary case: $-1 < v < 1$, is given, after a completely analogous procedure, by the expressions:

$$v(\zeta) = \frac{1 - v_2 + 2v_2 \operatorname{sn}^2 s_v}{v_2 - 1 + 2 \operatorname{sn}^2 s_v}$$

with

$$s_v(\zeta) = \frac{\pm \sqrt{-\frac{2\bar{\sigma}}{\sigma} \Omega}}{g_v} \zeta + s_{v_0}, \quad k_v^2 = \frac{2(v_2 - v_1)}{(v_2 - 1)(1 + v_1)}, \quad g_v = \frac{2}{\sqrt{(v_2 - 1)(1 + v_1)}}$$

Applying transformation (22) to these expressions of $u(\zeta)$ and $v(\zeta)$, and replacing the results in (5), a complete description in Cartesian coordinates of planetary orbits in the Northern hemisphere is obtained.

Analogously, all the integrals $I(u)$ and $I(v)$ solving equations (24) in the different ranges of u and v compatible with $\Omega < 0$ can be inverted by similar techniques. The ensuing analytic expressions are assembled in Appendix A. The $u(\zeta)$ and $v(\zeta)$ functions which respectively solve the u - and v -dynamics are smooth, bounded between turning points, and periodic with periods respectively $T_u \propto K(k_u^2)$ and $T_v \propto K(k_v^2)$, where $K(k^2)$ is the complete elliptic function of the first kind. The trajectories in all these cases are bounded between caustics in S_+^2 and dense, except if the u - and v -periods are commensurable.

- $\Omega > 0$. The procedure is more delicate in this case essentially because the trajectories complying with the ODE pair (24) reach the equator whereas there is admissible motion governed by (25) that also reach the equator coming from the Southern hemisphere. Therefore, it is convenient to investigate the inversion of the quadratures of the u -equations of both (24) and (25) in a global form. However, in the ‘‘angular’’ v -integrals there are no differences with respect to the $\Omega < 0$ range.

Let us focus on planetary orbits. An orbit of this type in S^2 is described by two pieces: the portion belonging to S^2_+ is a solution of equations (24) in the ranges:

$$u_1 < -1 < 1 < u_2 < u, \quad v_1 < -1 < v < 1 < v_2$$

whereas for the S^2_- piece we have equations (25) and ranges:

$$\tilde{u}_1 < -1 < 1 < \tilde{u}_2 < u, \quad \tilde{v}_1 < -1 < v < 1 < \tilde{v}_2$$

The first quadrature in (24) for the “radial” variable:

$$\pm \sqrt{\frac{2\bar{\sigma}}{\sigma}} \Omega \zeta = I(u) - I(u_0); \quad I(u) = \int_{u_2}^u \frac{dz}{\sqrt{(z^2 - 1)(z - u_1)(z - u_2)}}. \quad (41)$$

can be inverted with a change of variable like that explained before in the $\Omega < 0$ case. The solution is

$$u(\zeta) \equiv u(s_u) = \frac{u_2 - 1 + (u_2 + 1) \operatorname{dn}^2 s_u}{1 - u_2 + (u_2 + 1) \operatorname{dn}^2 s_u} \quad (42)$$

where:

$$s_u(\zeta) = \frac{\pm \sqrt{\frac{2\bar{\sigma}}{\sigma}} \Omega}{g_u} \zeta + s_{u_0}, \quad k_u^2 = \frac{2(u_2 - u_1)}{(1 - u_1)(1 + u_2)}, \quad g_u = \frac{2}{\sqrt{(1 - u_1)(1 + u_2)}}$$

A plot of $u(\zeta)$, see Figure 4 (left), shows several relevant features of $u(\zeta)$. First, the function (42) presents infinite poles, located at the points where: $\operatorname{dn}^2 s_u = \frac{u_2 - 1}{u_2 + 1}$. This is an expected result if one sees $u(\zeta)$ as a solution of the planar problem of two attractive centers with $h > 0$ re-interpreting ζ as the local time of this planar problem; the trajectory goes to infinity in a finite interval of the local time. However, in the sphere S^2 the sphero-conical variable $U(\zeta)$, given by (22), is bounded but exhibits finite discontinuities and reaches its maxima on the equator $U = 1$ at the poles of $u(\zeta)$, see Fig. 4(right). Second, it is remarkable, and a priori unexpected, that both $u(\zeta)$ and $U(\zeta)$ take negative values. The subtle interpretation of this fact is the understanding that, given the inversion problem posed by (41), its solution $u(\zeta) \equiv u(s_u)$ solves also the complementary problem: $y < u_1 < -1 < 1 < u_2$, i.e. the inversion problem of the elliptic integral:

$$I'(y) = \int_y^{u_1} \frac{dz}{\sqrt{(z^2 - 1)(z - u_1)(z - u_2)}}$$

in such a way that the inverse function $y(s)$ verifies: $y(s) = u(s_u + K)$, where $K = K(k_u^2)$. Thus, $u(s_u)$ defined in equation (42) represents simultaneously the genuine u -“radial” positive solution, $u \in (u_2, \infty)$, and the negative $y(s)$ -“radial” solution with $y \in (-\infty, u_1)$. Note that, according to the plot in Figure 4, these two solutions occur in consecutive intervals of the local time ζ .

A direct search for the solution of equation (25) in S^2_- , where $\tilde{u}_1 < -1 < 1 < \tilde{u}_2 < u$, requires the inversion of the elliptic integral in the next equation:

$$\pm \sqrt{\frac{2\bar{\sigma}}{\sigma}} \Omega \zeta = I'(u) - I'(u_0), \quad \tilde{I}(u) = \int_{\tilde{u}_2}^u \frac{dz}{\sqrt{(z^2 - 1)(z - \tilde{u}_1)(z - \tilde{u}_2)}}.$$

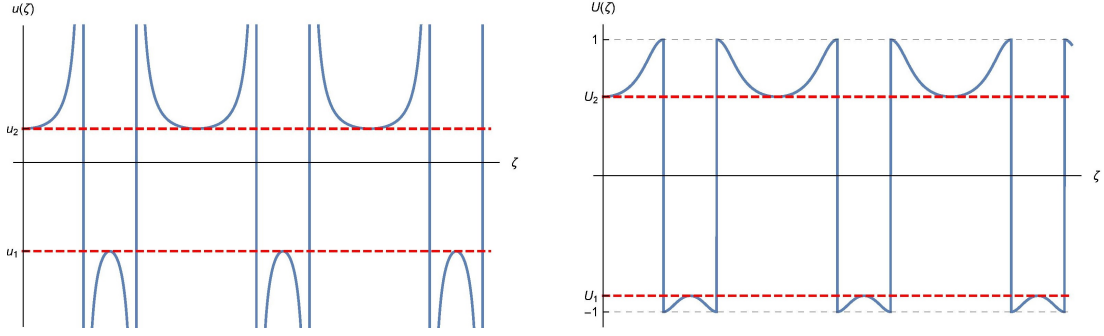


Figure 4: Graphics of the function $u(\zeta)$ defined in (42) and its partner $U(\zeta)$ in S^2 , corresponding to the values: $\Omega = \sqrt{3}$, $G = \frac{2\sqrt{3}}{3}$, $\sigma = \cos \frac{\pi}{6}$, $s_{u_0} = 0$.

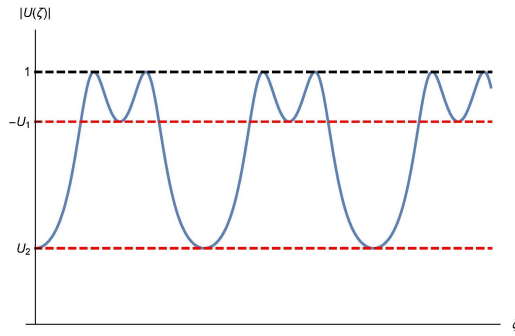


Figure 5: Graphics of the function $|U(\zeta)|$ corresponding to the values: $\Omega = \sqrt{3}$, $G = \frac{2\sqrt{3}}{3}$, $\sigma = \cos \frac{\pi}{6}$, $s_{u_0} = 0$.

Having in mind that $\tilde{u}_1 = -u_2$, $\tilde{u}_2 = -u_1$, we can write:

$$\tilde{I}(u) = \int_{-u_1}^u \frac{dz}{\sqrt{(z^2 - 1)(z + u_2)(z + u_1)}} = \int_{-u}^{u_1} \frac{dw}{\sqrt{(w^2 - 1)(w - u_2)(w - u_1)}} = I'(-u)$$

where the change of variable $z = -w$ has been performed. Thus, we conclude that the inversion of $\tilde{I}(u)$, i.e. the “radial” solution in S_-^2 , is tantamount to the inversion of $I'(-u)$ and consequently to minus the negative part of $u(s_u)$ given in (42). Therefore, we represent the “radial” solution simultaneously in both S_+^2 and S_-^2 by simply taking the absolute value $|u(\zeta)|$ of the solution given in (42). Moreover, with this identification the function $|U(\zeta)|$ is smooth, i.e. the gluing at the equator of the Northern and Southern branches of the orbits is continuous and differentiable, see Figure 5, with respect to the local time ζ .

This argument is valid also for the “radial” quadratures of the rest of different types of orbits that cross the equator. Thus, the general expression for the orbits in Cartesian coordinates over the sphere S^2 , using (22) in (5), can be written in a compact form valid for all the types of orbits described in

the previous Section as:

$$\begin{aligned}
X(\zeta) &= \frac{R\bar{\sigma} |u(\zeta)| v(\zeta)}{\sqrt{\bar{\sigma}^2 u^2(\zeta) + \sigma^2} \sqrt{\bar{\sigma}^2 v^2(\zeta) + \sigma^2}} \\
Y(\zeta) &= \frac{\pm R\sigma\bar{\sigma} \sqrt{u^2(\zeta) - 1} \sqrt{1 - v^2(\zeta)}}{\sqrt{\bar{\sigma}^2 u^2(\zeta) + \sigma^2} \sqrt{\bar{\sigma}^2 v^2(\zeta) + \sigma^2}} \\
Z(\zeta) &= \frac{R\sigma \operatorname{sg}[u(\zeta)]}{\sqrt{\bar{\sigma}^2 u^2(\zeta) + \sigma^2} \sqrt{\bar{\sigma}^2 v^2(\zeta) + \sigma^2}}.
\end{aligned} \tag{43}$$

Here, sg denotes the sign function and $(u(\zeta), v(\zeta))$ are the solutions of equations (20) or (21). Explicit expressions for (43) in all the different regimes are written in the Appendix.

The periodicity properties of the functions (43) are inherited from the Jacobi elliptic functions through the functions $u(\zeta)$ and $v(\zeta)$: Solutions (43) are products of periodic functions with different periods T_u and T_v . Consequently (43) will be periodic, and thus the orbits closed, only if T_u and T_v are commensurable, i.e. there exists $p, q \in \mathbb{N}^*$ such that:

$$pT_u = qT_v \tag{44}$$

otherwise the orbits will be dense inside the allowable region of S^2 .

The periods T_u and T_v are proportional to $K(k_u^2)$ and $K(k_v^2)$ respectively, with a factor that depends on the concrete Jacobi functions involved in the respective expressions of $u(\zeta)$ and $v(\zeta)$. The search for a closed orbit, having fixed the values of p , q and Ω (or G), requires to solve the transcendental equation (44) in the variable G (alternatively Ω). Explicit expressions for the periods, and concrete examples of closed orbits for different values of p and q are collected in the Appendix.

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Appendix. Explicit expressions for the different types of orbits

The set of parameters that determines the problem is R , θ_f , γ_1 and γ_2 , but after defining nondimensional variables the strengths can be measured with only one relative quantity: $\gamma = \frac{\gamma_2}{\gamma_1 + \gamma_2}$.

Our choice of integration constants to characterize the solutions (43), as functions of the nondimensional local time ζ introduced in (23), is: the two constants of motion Ω and G , and the two initial data s_{u_0} and s_{v_0} . Dependence in Ω and G is given implicitly through the values of the branching points:

$$\begin{aligned}
u_1 &= \frac{\sigma}{\bar{\sigma}} \left[\frac{-1}{2\Omega} - \sqrt{\frac{G}{\Omega} + \frac{1}{4\Omega^2}} \right], & u_2 &= \frac{\sigma}{\bar{\sigma}} \left[\frac{-1}{2\Omega} + \sqrt{\frac{G}{\Omega} + \frac{1}{4\Omega^2}} \right] \\
v_1 &= \frac{\sigma}{\bar{\sigma}} \left[\frac{(1-2\gamma)}{2\Omega} - \sqrt{\frac{G}{\Omega} + \frac{(1-2\gamma)^2}{4\Omega^2}} \right], & v_2 &= \frac{\sigma}{\bar{\sigma}} \left[\frac{(1-2\gamma)}{2\Omega} + \sqrt{\frac{G}{\Omega} + \frac{(1-2\gamma)^2}{4\Omega^2}} \right]
\end{aligned}$$

if $\Omega \neq 0$, and $u_1 = \frac{\sigma}{\bar{\sigma}}G$, $v_2 = \frac{\sigma}{\bar{\sigma}} \frac{-G}{(1-2\gamma)}$ for the $\Omega = 0$ case.

Remember also that the following notation have been introduced along the paper:

$$\sigma = \cos \theta_f, \quad \bar{\sigma} = \sin \theta_f; \quad \text{sn } s_u = \text{sn}(s_u(\zeta)|k_u^2)$$

and so on for the rest of Jacobi elliptic functions, where:

$$\begin{aligned} s_u \equiv s_u(\zeta) &= \frac{\pm \sqrt{\frac{2\bar{\sigma}}{\sigma}|\Omega|}}{g_u} \zeta + s_{u_0}, & s_v \equiv s_v(\zeta) &= \frac{\pm \sqrt{\frac{2\bar{\sigma}}{\sigma}|\Omega|}}{g_v} \zeta + s_{v_0} & \text{if } \Omega \neq 0 \\ s_u \equiv s_u(\zeta) &= \frac{\pm \sqrt{2}}{g_u} \zeta + s_{u_0}, & s_v \equiv s_v(\zeta) &= \frac{\pm \sqrt{2}}{g_v} \zeta + s_{v_0} & \text{if } \Omega = 0 \end{aligned}$$

in such a way that initial conditions are: $s_{u_0} = s_u(0)$ and $s_{v_0} = s_v(0)$.

With all these considerations, the orbits for the two fixed centers problem in S^2 are:

$$\Omega > 0: \text{ Orbits that cross the equator.}$$

- Planetary orbits- t_p , see (33):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_2 - (u_2 + 1) \text{dn}^2 s_u) (1 - v_1 + 2v_1 \text{sn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} \sqrt{u_2^2 - 1} \sqrt{v_1^2 - 1} \text{dn} s_u \text{sn} s_v \text{cn} s_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (u_2 - 1 - (u_2 + 1) \text{dn}^2 s_u) (v_1 - 1 + 2 \text{sn}^2 s_v) \end{cases} \quad (45)$$

where

$$\begin{aligned} \Upsilon_u &= \sqrt{(u_2 - 1)^2 - 2(u_2^2 - 1)(\sigma^2 - \bar{\sigma}^2) \text{dn}^2 s_u + (u_2 + 1)^2 \text{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{(v_1 - 1)^2 + 4(1 - v_1)(\bar{\sigma}^2 v_1 - \sigma^2) \text{sn}^2 s_v + 4(\bar{\sigma}^2 v_1^2 + \sigma^2) \text{sn}^4 s_v} \end{aligned}$$

$$k_u^2 = \frac{2(u_2 - u_1)}{(1 - u_1)(1 + u_2)}, \quad g_u = \frac{2}{\sqrt{(1 - u_1)(1 + u_2)}}, \quad k_v^2 = \frac{2(v_2 - v_1)}{(1 - v_1)(1 + v_2)}, \quad g_v = \frac{2}{\sqrt{(1 - v_1)(1 + v_2)}}$$

- Lemniscatic orbits- t_l (34):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (u_2 - 1 - 2u_2 \text{dn}^2 s_u) (1 - v_1 + 2v_1 \text{sn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} k_u \sqrt{1 - u_2^2} \sqrt{v_1^2 - 1} \text{dn} s_u \text{sn} s_u \text{sn} s_v \text{cn} s_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (1 - u_2 - 2 \text{dn}^2 s_u) (v_1 - 1 + 2 \text{sn}^2 s_v) \end{cases} \quad (46)$$

$$\begin{aligned} \Upsilon_u &= \sqrt{(u_2 - 1)^2 + 4(1 - u_2)(\bar{\sigma}^2 u_2 - \sigma^2) \text{dn}^2 s_u + 4(\bar{\sigma}^2 u_2^2 + \sigma^2) \text{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{(v_1 - 1)^2 + 4(1 - v_1)(\bar{\sigma}^2 v_1 - \sigma^2) \text{sn}^2 s_v + 4(\bar{\sigma}^2 v_1^2 + \sigma^2) \text{sn}^4 s_v} \end{aligned}$$

$$k_u^2 = \frac{(1-u_1)(1+u_2)}{2(u_2-u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2-u_1)}}, \quad k_v^2 = \frac{2(v_2-v_1)}{(1-v_1)(1+v_2)}, \quad g_v = \frac{2}{\sqrt{(1-v_1)(1+v_2)}}$$

- Satellitary orbits- $t_{s'}$ (35):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1-u_2 + 2u_2 \operatorname{dn}^2 s_u) (2v_1 + (1-v_1) \operatorname{sn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} k_u \sqrt{1-u_2^2} \sqrt{1-v_1^2} \operatorname{dn} s_u \operatorname{sn} s_u \operatorname{cn} s_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (u_2 - 1 + 2 \operatorname{dn}^2 s_u) (2 - (1-v_1) \operatorname{sn}^2 s_v) \end{cases} \quad (47)$$

$$\begin{aligned} \Upsilon_u &= \sqrt{(u_2-1)^2 + 4(1-u_2)(\bar{\sigma}^2 u_2 - \sigma^2) \operatorname{dn}^2 s_u + 4(\bar{\sigma}^2 u_2^2 + \sigma^2) \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{4(\bar{\sigma}^2 v_1^2 + \sigma^2) + 4(1-v_1)(\bar{\sigma}^2 v_1 - \sigma^2) \operatorname{sn}^2 s_v + (v_1-1)^2 \operatorname{sn}^4 s_v} \end{aligned}$$

$$k_u^2 = \frac{(1-u_1)(1+u_2)}{2(u_2-u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2-u_1)}}, \quad k_v^2 = \frac{(1-v_1)(1+v_2)}{2(v_2-v_1)}, \quad g_v = \frac{\sqrt{2}}{\sqrt{(v_2-v_1)}}$$

- Dual Satellitary orbits- t_{ds} (36):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1-u_2 + 2u_2 \operatorname{dn}^2 s_u) (1+v_1 - (1-v_1) \operatorname{dn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} k_u \sqrt{1-u_2^2} \sqrt{1-v_1^2} \operatorname{dn} s_u \operatorname{sn} s_u \operatorname{dn} s_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (u_2 - 1 + 2 \operatorname{dn}^2 s_u) (1+v_1 + (1-v_1) \operatorname{dn}^2 s_v) \end{cases} \quad (48)$$

$$\begin{aligned} \Upsilon_u &= \sqrt{(u_2-1)^2 + 4(1-u_2)(\bar{\sigma}^2 u_2 - \sigma^2) \operatorname{dn}^2 s_u + 4(\bar{\sigma}^2 u_2^2 + \sigma^2) \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{(1+v_1)^2 + 2(1-v_1^2)(\sigma^2 - \bar{\sigma}^2) \operatorname{dn}^2 s_v + (1-v_1)^2 \operatorname{dn}^4 s_v} \end{aligned}$$

$$k_u^2 = \frac{(1-u_1)(1+u_2)}{2(u_2-u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2-u_1)}}, \quad k_v^2 = \frac{2(v_2-v_1)}{(1-v_1)(1+v_2)}, \quad g_v = \frac{2}{\sqrt{(1-v_1)(1+v_2)}}$$

- Meridian Planetary orbits- t_{mp} (37):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (u_2 + 1 - 2u_2 \operatorname{sn}^2 s_u) (1+v_1 - (1-v_1) \operatorname{dn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} \sqrt{1-u_2^2} \sqrt{1-v_1^2} \operatorname{cn} s_u \operatorname{sn} s_u \operatorname{dn} s_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (1+u_2 - 2 \operatorname{sn}^2 s_u) (1+v_1 + (1-v_1) \operatorname{dn}^2 s_v) \end{cases} \quad (49)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(1+u_2)^2 - 4(1+u_2)(\bar{\sigma}^2 u_2 + \sigma^2) \operatorname{sn}^2 s_u + 4(\bar{\sigma}^2 u_2^2 + \sigma^2) \operatorname{sn}^4 s_u} \\ \Upsilon_v &= \sqrt{(1+v_1)^2 + 2(1-v_1^2)(\sigma^2 - \bar{\sigma}^2) \operatorname{dn}^2 s_v + (1-v_1)^2 \operatorname{dn}^4 s_v}\end{aligned}$$

$$k_u^2 = \frac{2(u_2 - u_1)}{(1-u_1)(1+u_2)}, \quad g_u = \frac{2}{\sqrt{(1-u_1)(1+u_2)}}, \quad k_v^2 = \frac{2(v_2 - v_1)}{(1-v_1)(1+v_2)}, \quad g_v = \frac{2}{\sqrt{(1-v_1)(1+v_2)}}$$

• Having into account the involved Jacobi functions in each type of solutions, the u - and v - periods for the different orbits with $\Omega > 0$ are:

$$\begin{aligned}t_p \text{ orbits : } \quad T_u &= \frac{g_u}{\sqrt{\frac{2\bar{\sigma}}{\sigma}\Omega}} 2K(k_u^2) \quad , \quad T_v = \frac{g_v}{\sqrt{\frac{2\bar{\sigma}}{\sigma}\Omega}} 4K(k_v^2) \\ t_l \text{ and } t_{s'} \text{ orbits : } \quad T_u &= \frac{g_u}{\sqrt{\frac{2\bar{\sigma}}{\sigma}\Omega}} 4K(k_u^2) \quad , \quad T_v = \frac{g_v}{\sqrt{\frac{2\bar{\sigma}}{\sigma}\Omega}} 4K(k_v^2) \\ t_{ds} \text{ and } t_{mp} \text{ orbits : } \quad T_u &= \frac{g_u}{\sqrt{\frac{2\bar{\sigma}}{\sigma}\Omega}} 4K(k_u^2) \quad , \quad T_v = \frac{g_v}{\sqrt{\frac{2\bar{\sigma}}{\sigma}\Omega}} 2K(k_v^2)\end{aligned}$$

$\Omega < 0$: Orbits that lie only in the Northern hemisphere.

• Planetary orbits- t_p of type 1, (26):

$$\left\{ \begin{aligned} X(\zeta) &= \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (u_1 - 1 + (u_1 + 1) \operatorname{dn}^2 s_u) (|1 + v_1| (1 - \operatorname{cns}_v) - |1 - v_1| (1 + \operatorname{cns}_v)) \\ Y(\zeta) &= \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} \sqrt{u_1^2 - 1} \sqrt{|1 - v_1| |1 + v_1|} \operatorname{dnc}_u \operatorname{sn}_v \\ Z(\zeta) &= \frac{R}{\Upsilon_u \Upsilon_v} \sigma (1 - u_1 + (u_1 + 1) \operatorname{dn}^2 s_u) (|1 + v_1| (1 - \operatorname{cns}_v) + |1 - v_1| (1 + \operatorname{cns}_v)) \end{aligned} \right. \quad (50)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(u_1 - 1)^2 - 2(u_1^2 - 1)(\sigma^2 - \bar{\sigma}^2) \operatorname{dn}^2 s_u + (u_1 + 1)^2 \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{|1 - v_1|^2 (1 + \operatorname{cns}_v)^2 + 2|1 - v_1| |1 + v_1| (\sigma^2 - \bar{\sigma}^2) \operatorname{sn}^2 s_v + |1 + v_1|^2 (1 - \operatorname{cns}_v)^2}\end{aligned}$$

$$k_u^2 = \frac{2(u_2 - u_1)}{(u_1 + 1)(u_2 - 1)}, \quad g_u = \frac{2}{\sqrt{(u_1 + 1)(u_2 - 1)}}, \quad k_v^2 = \frac{4 - (|1 - v_1| - |1 + v_1|)^2}{4|1 - v_1| |1 + v_1|}, \quad g_v = \frac{1}{\sqrt{|1 - v_1| |1 + v_1|}}$$

• Planetary orbits- t_p of type 2, (27):

$$\left\{ \begin{aligned} X(\zeta) &= \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_1 - (u_1 + 1) \operatorname{dn}^2 s_u) (1 - v_2 + 2v_2 \operatorname{sn}^2 s_v) \\ Y(\zeta) &= \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} \sqrt{u_1^2 - 1} \sqrt{v_2^2 - 1} \operatorname{dnc}_u \operatorname{sn}_v \operatorname{cns}_v \\ Z(\zeta) &= \frac{R}{\Upsilon_u \Upsilon_v} \sigma (-1 + u_1 - (u_1 + 1) \operatorname{dn}^2 s_u) (v_2 - 1 + 2 \operatorname{sn}^2 s_v) \end{aligned} \right. \quad (51)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(u_1 - 1)^2 - 2(u_1^2 - 1)(\sigma^2 - \bar{\sigma}^2) \operatorname{dn}^2 s_u + (u_1 + 1)^2 \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{(v_2 - 1)^2 + 4(1 - v_2)(\bar{\sigma}^2 v_2 - \sigma^2) \operatorname{sn}^2 s_v + 4(\bar{\sigma}^2 v_2^2 + \sigma^2) \operatorname{sn}^4 s_v}\end{aligned}$$

$$k_u^2 = \frac{2(u_2 - u_1)}{(u_1 + 1)(u_2 - 1)}, \quad g_u = \frac{2}{\sqrt{(u_1 + 1)(u_2 - 1)}}, \quad k_v^2 = \frac{2(v_2 - v_1)}{(v_1 + 1)(v_2 - 1)}, \quad g_v = \frac{2}{\sqrt{(v_1 + 1)(v_2 - 1)}}$$

- Lemniscatic orbits- t_l of type 1, (28):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_1 + 2u_1 \operatorname{dn}^2 s_u) (|1 + v_1| (1 - \operatorname{cns}_v) - |1 - v_1| (1 + \operatorname{cns}_v)) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} k_u \sqrt{1 - u_1^2} \sqrt{|1 - v_1| |1 + v_1|} \operatorname{dns}_u \operatorname{sns}_u \operatorname{sns}_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (u_1 - 1 + 2 \operatorname{dn}^2 s_u) (|1 + v_1| (1 - \operatorname{cns}_v) + |1 - v_1| (1 + \operatorname{cns}_v)) \end{cases} \quad (52)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(u_1 - 1)^2 + 4(1 - u_1)(\bar{\sigma}^2 u_1 - \sigma^2) \operatorname{dn}^2 s_u + 4(\bar{\sigma}^2 u_1^2 + \sigma^2) \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{|1 - v_1|^2 (1 + \operatorname{cns}_v)^2 + 2|1 - v_1| |1 + v_1| (\sigma^2 - \bar{\sigma}^2) \operatorname{sn}^2 s_v + |1 + v_1|^2 (1 - \operatorname{cns}_v)^2}\end{aligned}$$

$$k_u^2 = \frac{(u_1 + 1)(u_2 - 1)}{2(u_2 - u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2 - u_1)}}, \quad k_v^2 = \frac{4 - (|1 - v_1| - |1 + v_1|)^2}{4|1 - v_1| |1 + v_1|}, \quad g_v = \frac{1}{\sqrt{|1 - v_1| |1 + v_1|}}$$

- Lemniscatic orbits- t_l of type 2, (29):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (u_1 - 1 - 2u_1 \operatorname{dn}^2 s_u) (1 - v_2 + 2v_2 \operatorname{sn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} k_u \sqrt{1 - u_1^2} \sqrt{v_2^2 - 1} \operatorname{dns}_u \operatorname{sns}_u \operatorname{sns}_v \operatorname{cns}_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (1 - u_1 - 2 \operatorname{dn}^2 s_u) (v_2 - 1 + 2 \operatorname{sn}^2 s_v) \end{cases} \quad (53)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(u_1 - 1)^2 + 4(1 - u_1)(\bar{\sigma}^2 u_1 - \sigma^2) \operatorname{dn}^2 s_u + 4(\bar{\sigma}^2 u_1^2 + \sigma^2) \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{(v_2 - 1)^2 + 4(1 - v_2)(\bar{\sigma}^2 v_2 - \sigma^2) \operatorname{sn}^2 s_v + 4(\bar{\sigma}^2 v_2^2 + \sigma^2) \operatorname{sn}^4 s_v}\end{aligned}$$

$$k_u^2 = \frac{(u_1 + 1)(u_2 - 1)}{2(u_2 - u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2 - u_1)}}, \quad k_v^2 = \frac{2(v_2 - v_1)}{(v_1 + 1)(v_2 - 1)}, \quad g_v = \frac{2}{\sqrt{(v_1 + 1)(v_2 - 1)}}$$

- Satellitary orbits- t_s in zone 1, (30):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_1 + 2u_1 \operatorname{dn}^2 s_u) (v_2(1 - v_1) + v_1(v_2 - 1) \operatorname{sn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 2\sigma \bar{\sigma} k_u \sqrt{1 - u_1^2} \sqrt{1 - v_2^2} (1 - v_1) \operatorname{dns}_u \operatorname{sns}_u \operatorname{dns}_v \operatorname{cns}_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (u_1 - 1 + 2 \operatorname{dn}^2 s_u) (1 - v_1 - (1 - v_2) \operatorname{sn}^2 s_v) \end{cases} \quad (54)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(u_1 - 1)^2 + 4(1 - u_1)(\bar{\sigma}^2 u_1 - \sigma^2) \operatorname{dn}^2 s_u + 4(\bar{\sigma}^2 u_1^2 + \sigma^2) \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{(v_1 - 1)^2(\bar{\sigma}^2 v_2^2 + \sigma^2) - 2(1 - v_1)(1 - v_2)(\bar{\sigma}^2 v_1 v_2 + \sigma^2) \operatorname{sn}^2 s_v + (v_2 - 1)^2(\bar{\sigma}^2 v_1^2 + \sigma^2) \operatorname{sn}^4 s_v}\end{aligned}$$

$$k_u^2 = \frac{(u_1 + 1)(u_2 - 1)}{2(u_2 - u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2 - u_1)}}, \quad k_v^2 = \frac{(1 + v_1)(1 - v_2)}{(1 - v_1)(1 + v_2)}, \quad g_v = \frac{2}{\sqrt{(1 - v_1)(1 + v_2)}}$$

- Satellites orbits- t_s in zone 2, (31):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_1 + 2u_1 \operatorname{dn}^2 s_u) (2v_2 - (1 + v_2) \operatorname{dn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} k_u k_v \sqrt{1 - u_1^2} \sqrt{1 - v_2^2} \operatorname{dn} s_u \operatorname{sn} s_u \operatorname{sn} s_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (u_1 - 1 + 2 \operatorname{dn}^2 s_u) (2 - (1 + v_2) \operatorname{dn}^2 s_v) \end{cases} \quad (55)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(u_1 - 1)^2 + 4(1 - u_1)(\bar{\sigma}^2 u_1 - \sigma^2) \operatorname{dn}^2 s_u + 4(\bar{\sigma}^2 u_1^2 + \sigma^2) \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{4(\bar{\sigma}^2 v_2^2 + \sigma^2) - 4(1 + v_2)(\bar{\sigma}^2 v_2 + \sigma^2) \operatorname{dn}^2 s_v + (1 + v_2)^2 \operatorname{dn}^4 s_v}\end{aligned}$$

$$k_u^2 = \frac{(u_1 + 1)(u_2 - 1)}{2(u_2 - u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2 - u_1)}}, \quad k_v^2 = \frac{(1 + v_1)(1 - v_2)}{(1 - v_1)(1 + v_2)}, \quad g_v = \frac{2}{\sqrt{(1 - v_1)(1 + v_2)}}$$

- Satellites orbits- $t_{s'}$ (32):

$$\begin{cases} X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_1 + 2u_1 \operatorname{dn}^2 s_u) (2v_2 + (1 - v_2) \operatorname{sn}^2 s_v) \\ Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 4\sigma \bar{\sigma} k_u \sqrt{1 - u_1^2} \sqrt{1 - v_2^2} \operatorname{dn} s_u \operatorname{sn} s_u \operatorname{cn} s_v \\ Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma (u_1 - 1 + 2 \operatorname{dn}^2 s_u) (2 - (1 - v_2) \operatorname{sn}^2 s_v) \end{cases} \quad (56)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{(u_1 - 1)^2 + 4(1 - u_1)(\bar{\sigma}^2 u_1 - \sigma^2) \operatorname{dn}^2 s_u + 4(\bar{\sigma}^2 u_1^2 + \sigma^2) \operatorname{dn}^4 s_u} \\ \Upsilon_v &= \sqrt{4(\bar{\sigma}^2 v_2^2 + \sigma^2) + 4(1 - v_2)(\bar{\sigma}^2 v_2 - \sigma^2) \operatorname{sn}^2 s_v + (v_2 - 1)^2 \operatorname{sn}^4 s_v}\end{aligned}$$

$$k_u^2 = \frac{(u_1 + 1)(u_2 - 1)}{2(u_2 - u_1)}, \quad g_u = \frac{\sqrt{2}}{\sqrt{(u_2 - u_1)}}, \quad k_v^2 = \frac{(v_1 + 1)(v_2 - 1)}{2(v_2 - v_1)}, \quad g_v = \frac{\sqrt{2}}{\sqrt{(v_2 - v_1)}}$$

- The u - and v - periods in the case $\Omega < 0$ are:

$$\begin{aligned}
t_p(1), t_p(2) \text{ orbits : } \quad T_u &= \frac{g_u}{\sqrt{-\frac{2\bar{\sigma}}{\sigma}\Omega}} 2K(k_u^2) \quad , \quad T_v = \frac{g_v}{\sqrt{-\frac{2\bar{\sigma}}{\sigma}\Omega}} 4K(k_v^2) \\
t_l(1), t_l(2), t_s(1), t_s(2) \text{ and } t_{s'} \text{ orbits : } T_u &= \frac{g_u}{\sqrt{-\frac{2\bar{\sigma}}{\sigma}\Omega}} 4K(k_u^2) \quad , \quad T_v = \frac{g_v}{\sqrt{-\frac{2\bar{\sigma}}{\sigma}\Omega}} 4K(k_v^2)
\end{aligned}$$

$\Omega = 0$: Orbits that lie in the Northern hemisphere bounded by the equator.

- Planetary orbits- t_p , in this case: $-1 < 1 < u_1 < u$, $v_2 < -1 < v < 1$.

$$\begin{cases}
X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (u_1 - \text{sn}^2 s_u) (-1 - v_2 + v_2 \text{dn}^2 s_v) \\
Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 2\sigma \bar{\sigma} \sqrt{u_1^2 - 1} \sqrt{\frac{1+v_2}{v_2-1}} \text{dn} s_u \text{sn} s_v \text{cn} s_v \\
Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma \text{cn}^2 s_u \text{dn}^2 s_v
\end{cases} \quad (57)$$

$$\begin{aligned}
\Upsilon_u &= \sqrt{(\bar{\sigma}^2 u_1^2 + \sigma^2) - 2(\bar{\sigma}^2 u_1 + \sigma^2) \text{sn}^2 s_u + \text{sn}^4 s_u} \\
\Upsilon_v &= \sqrt{\bar{\sigma}^2 (1 + v_2)^2 - 2\bar{\sigma}^2 v_2 (1 + v_2) \text{dn}^2 s_v + (\bar{\sigma}^2 v_2^2 + \sigma^2) \text{dn}^4 s_v}
\end{aligned}$$

$$k_u^2 = \frac{2}{(u_1 + 1)} \quad , \quad g_u = \frac{2}{\sqrt{(1 + u_1)}} \quad , \quad k_v^2 = \frac{2}{(1 - v_2)} \quad , \quad g_v = \frac{2}{\sqrt{(1 - v_2)}}$$

- Lemniscatic orbits- t_l : $-1 < u_1 < 1 < u$, $v_2 < -1 < v < 1$.

$$\begin{cases}
X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_1 \text{sn}^2 s_u) (-1 - v_2 + v_2 \text{dn}^2 s_v) \\
Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} 2\sqrt{2} \sigma \bar{\sigma} \sqrt{1 - u_1} \sqrt{\frac{1+v_2}{v_2-1}} \text{dn} s_u \text{sn} s_u \text{sn} s_v \text{cn} s_v \\
Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma \text{cn}^2 s_u \text{dn}^2 s_v
\end{cases} \quad (58)$$

$$\begin{aligned}
\Upsilon_u &= \sqrt{1 - 2(\bar{\sigma}^2 u_1 + \sigma^2) \text{sn}^2 s_u + (\bar{\sigma}^2 u_1^2 + \sigma^2) \text{sn}^4 s_u} \\
\Upsilon_v &= \sqrt{\bar{\sigma}^2 (1 + v_2)^2 - 2\bar{\sigma}^2 v_2 (1 + v_2) \text{dn}^2 s_v + (\bar{\sigma}^2 v_2^2 + \sigma^2) \text{dn}^4 s_v}
\end{aligned}$$

$$k_u^2 = \frac{(u_1 + 1)}{2} \quad , \quad g_u = \sqrt{2} \quad , \quad k_v^2 = \frac{2}{(1 - v_2)} \quad , \quad g_v = \frac{2}{\sqrt{(1 - v_2)}}$$

- Satellitary orbits- $t_{s'}$: $-1 < u_1 < 1 < u$, $-1 < v_2 < v < 1$.

$$\begin{cases}
X(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \bar{\sigma} (1 - u_1 \text{sn}^2 s_u) (1 + v_2 - \text{dn}^2 s_v) \\
Y(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sqrt{2} \sigma \bar{\sigma} \sqrt{1 - u_1} \sqrt{1 - v_2^2} \text{dn} s_u \text{sn} s_u \text{cn} s_v \\
Z(\zeta) = \frac{R}{\Upsilon_u \Upsilon_v} \sigma \text{cn}^2 s_u \text{dn}^2 s_v
\end{cases} \quad (59)$$

$$\begin{aligned}\Upsilon_u &= \sqrt{1 - 2(\bar{\sigma}^2 u_1 + \sigma^2) \operatorname{sn}^2 s_u + (\bar{\sigma}^2 u_1^2 + \sigma^2) \operatorname{sn}^4 s_u} \\ \Upsilon_v &= \sqrt{\bar{\sigma}^2(1 + v_2)^2 - 2\bar{\sigma}^2(1 + v_2) \operatorname{dn}^2 s_v + \operatorname{dn}^4 s_v}\end{aligned}$$

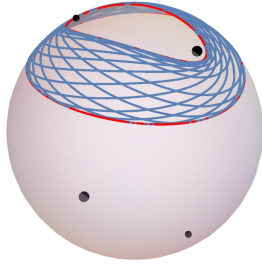
$$k_u^2 = \frac{(u_1 + 1)}{2} \quad , \quad g_u = \sqrt{2} \quad , \quad k_v^2 = \frac{(1 - v_2)}{2} \quad , \quad g_v = \sqrt{2}$$

- Finally, the u - and v - periods for these orbits with $\Omega = 0$ are:

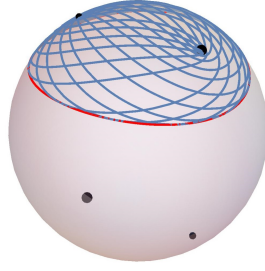
$$\begin{aligned}t_p \text{ orbits :} \quad & T_u = \frac{g_u}{\sqrt{2}} 2K(k_u^2) \quad , \quad T_v = \frac{g_v}{\sqrt{2}} 4K(k_v^2) \\ t_l \text{ and } t_{s'} \text{ orbits :} \quad & T_u = \frac{g_u}{\sqrt{2}} 4K(k_u^2) \quad , \quad T_v = \frac{g_v}{\sqrt{2}} 4K(k_v^2)\end{aligned}$$

In Figure 6 there are represented several orbits with $R = 1$, $\gamma = \frac{1}{3}$ and $\theta_f = \frac{\pi}{6}$, one for each different situation. These orbits are dense in all the cases and they are depicted in the interval $\zeta \in [0, 70]$.

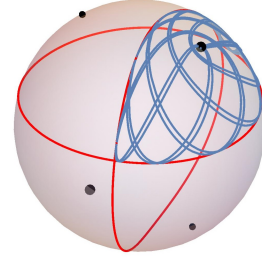
We can see in Figure 7 several closed orbits of different types, with specification of the values of p , q and initial data s_{u_0} and s_{v_0} . In all the cases p , q and the constant of motion Ω have been fixed, thus G has been calculated by solving numerically equation (44).



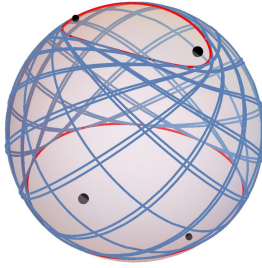
(a) $t_p : \frac{\bar{\sigma}}{\sigma}\Omega = -0.27, \frac{\sigma}{\sigma}G = 0.8$
 $s_{u_0} = 0, s_{v_0} = 0$



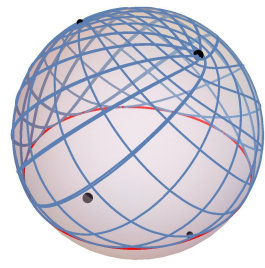
(b) $t_l : \frac{\bar{\sigma}}{\sigma}\Omega = -0.3, \frac{\sigma}{\sigma}G = 0.6$
 $s_{u_0} = 1, s_{v_0} = 0$



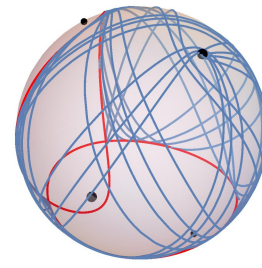
(c) $t_{s'} : \frac{\bar{\sigma}}{\sigma}\Omega = -0.2, \frac{\sigma}{\sigma}G = -0.1$
 $s_{u_0} = 1, s_{v_0} = 0$



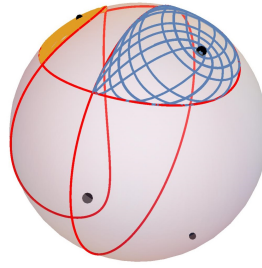
(d) $t_p : \frac{\bar{\sigma}}{\sigma}\Omega = 0.5, \frac{\sigma}{\sigma}G = 2$
 $s_{u_0} = 1, s_{v_0} = 2$



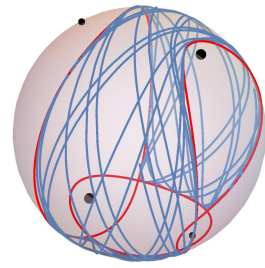
(e) $t_l : \frac{\bar{\sigma}}{\sigma}\Omega = 0.25, \frac{\sigma}{\sigma}G = 1$
 $s_{u_0} = 0, s_{v_0} = 0$



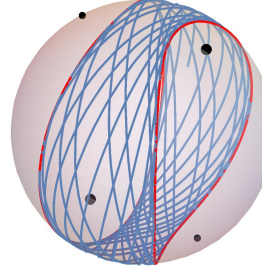
(f) $t_{s'} : \frac{\bar{\sigma}}{\sigma}\Omega = 0.5, \frac{\sigma}{\sigma}G = 0.5$
 $s_{u_0} = 1, s_{v_0} = 2$



(g) $t_s : \frac{\bar{\sigma}}{\sigma}\Omega = -0.5, \frac{\sigma}{\sigma}G = 0$
 $s_{u_0} = 1, s_{v_0} = 0$

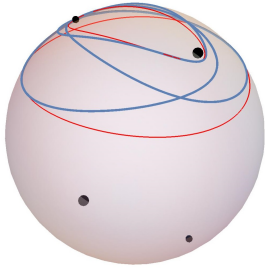


(h) $t_{ds} : \frac{\bar{\sigma}}{\sigma}\Omega = 0.8, \frac{\sigma}{\sigma}G = 0.2$
 $s_{u_0} = 1, s_{v_0} = 2$

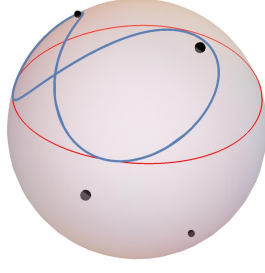


(i) $t_{mp} : \frac{\bar{\sigma}}{\sigma}\Omega = 1.5, \frac{\sigma}{\sigma}G = 0.2$
 $s_{u_0} = 1, s_{v_0} = 2$

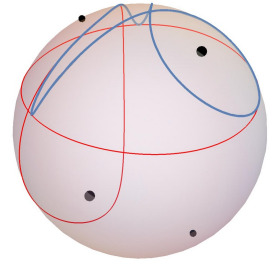
Figure 6: Orbits in S^2 . In all cases: $\gamma = \frac{1}{3}, \sigma = \cos \frac{\pi}{6}, \bar{\sigma} = \sin \frac{\pi}{6}$.



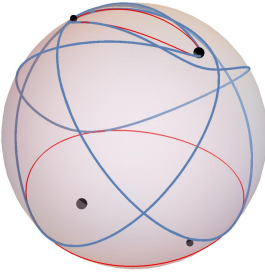
(a) $t_p : \frac{\bar{\sigma}}{\sigma}\Omega = -0.25, \frac{\sigma}{\bar{\sigma}}G \cong 0.80727$
 $2T_u = 3T_v, s_{u_0} = 0, s_{v_0} = 0$



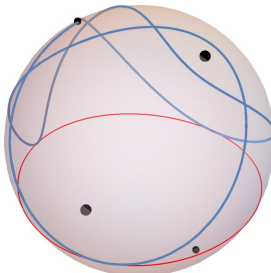
(b) $t_l : \frac{\bar{\sigma}}{\sigma}\Omega = -1/5, \frac{\sigma}{\bar{\sigma}}G \cong 0.29835$
 $T_u = T_v, s_{u_0} = 3, s_{v_0} = 0$



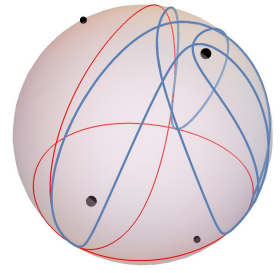
(c) $t_{s'} : \frac{\bar{\sigma}}{\sigma}\Omega = -0.25, \frac{\sigma}{\bar{\sigma}}G \cong 0.10725$
 $3T_u = T_v, s_{u_0} = 3, s_{v_0} = -1$



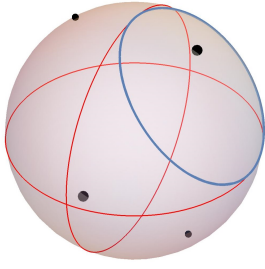
(d) $t_p : \frac{\bar{\sigma}}{\sigma}\Omega = 0.5, \frac{\sigma}{\bar{\sigma}}G \cong 1.56826$
 $3T_u = 2T_v, s_{u_0} = 1, s_{v_0} = 0$



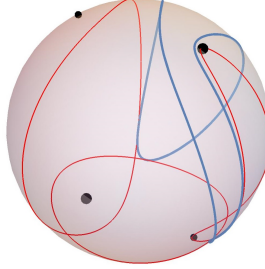
(e) $t_l : \frac{\bar{\sigma}}{\sigma}\Omega = 0.25, \frac{\sigma}{\bar{\sigma}}G \cong 0.72393$
 $3T_u = 2T_v, s_{u_0} = 0, s_{v_0} = 0$



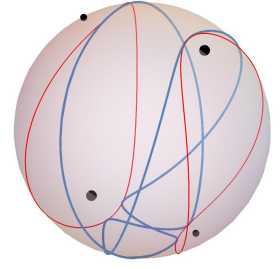
(f) $t_{s'} : \frac{\bar{\sigma}}{\sigma}\Omega = 0.3, \frac{\sigma}{\bar{\sigma}}G \cong 0.07292$
 $5T_u = 3T_v, s_{u_0} = 3, s_{v_0} = 1$



(g) $t_{s'} : \frac{\bar{\sigma}}{\sigma}\Omega = 0, \frac{\sigma}{\bar{\sigma}}G = 0$
 $T_u = T_v, s_{u_0} = 0, s_{v_0} = 0$



(h) $t_{ds} : \frac{\bar{\sigma}}{\sigma}\Omega = 0.6, \frac{\sigma}{\bar{\sigma}}G \cong 0.23559$
 $T_u = T_v, s_{u_0} = 3, s_{v_0} = 0$



(i) $t_{mp} : \frac{\bar{\sigma}}{\sigma}\Omega = 1.5, \frac{\sigma}{\bar{\sigma}}G \cong 0.47580$
 $T_u = 3T_v, s_{u_0} = 0, s_{v_0} = 0$

Figure 7: Closed orbits in S^2 . In all cases: $\gamma = \frac{1}{3}, \sigma = \cos \frac{\pi}{6}, \bar{\sigma} = \sin \frac{\pi}{6}$.

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